Numerical Optimal Control with DAEs Lecture 11: High-Index DAEs

Sébastien Gros

AWESCO PhD course

- Why are DAEs not always "easy" to solve ?
- What is a DAE index ? How does it make the DAE "easy" or not ?
- What to do about it ?
- What happens with DAE models for mechanical systems ?
- Some possible additional numerical problems

Outline

1 "Easy" & "Hard" DAEs

2 Differential Index

B Index Reduction

4) Constraints drift

Optimal Control with DAEs, lecture 12

 22^{nd} of February, 2016

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DAE - Easy & Hard DAEs

DAE:
$\mathbf{F}\left(\dot{\mathbf{x}},\mathbf{x},\mathbf{z},\mathbf{u}\right)=0$
"At any time instant, for given x, u, the DAE equation provides x, z, generating the trajectories."

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"At any time instant, for
given \mathbf{x}, \mathbf{u} , the DAE equation provides $\dot{\mathbf{x}}, \mathbf{z}$,
generating the
trajectories."

What is a "well-behaved" DAE ?

$$\mathbf{F}\left(\dot{\mathbf{x}},\mathbf{x},\mathbf{z},\mathbf{u}\right)=\mathbf{0}$$

... i.e. when can we compute $\dot{\mathbf{x}}, \mathbf{z}$ "easily" ?

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 $\begin{array}{l} \mbox{Consider } {\bf F}\left({\dot {\bf x}}, {\bf x}, {\bf z}, {\bf u} \right) = 0 \mbox{ as a root-finding problem in } {\dot {\bf x}}, \, {\bf z}. \mbox{ When can we find } {\dot {\bf x}}, \, {\bf z}, \\ \mbox{ e.g. using Newton } ? \end{array}$

Consider the DAE:

 $\mathbf{F}\left(\dot{\mathbf{x}},\mathbf{z},\mathbf{x},\mathbf{u}\right)=\mathbf{0}$

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Consider the DAE:

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If the matrix:

$$\begin{bmatrix} \frac{\partial \mathbf{F}}{\partial \dot{\mathbf{x}}} & \frac{\partial \mathbf{F}}{\partial \mathbf{z}} \end{bmatrix}$$

is full-rank at \mathbf{x}, \mathbf{u}

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 $\dot{\mathbf{x}}, \mathbf{z} = \boldsymbol{\xi}(\mathbf{x}, \mathbf{u})$

such that:

 $\mathbf{F}(\boldsymbol{\xi}(\mathbf{x},\mathbf{u}),\mathbf{x},\mathbf{u}) = \mathbf{0}$

holds in a neighborhood of \mathbf{x}, \mathbf{u} .

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Proof: from implicit function theorem.

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Consequence:

 $\left[\begin{array}{cc} \frac{\partial \mathbf{F}}{\partial \dot{\mathbf{x}}} & \frac{\partial \mathbf{F}}{\partial \mathbf{z}} \end{array}\right]$

full-rank at \mathbf{x}, \mathbf{u} guarantees that the DAE is "solvable" at $\dot{\mathbf{x}}, \mathbf{z}$

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Classical numerical methods can treat "easy" DAEs

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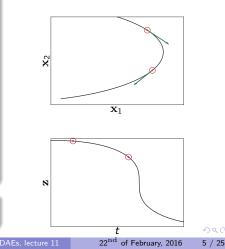
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Semi-explicit DAE:

$$\tilde{\mathbf{F}} = \begin{bmatrix} \dot{\mathbf{x}} - \mathbf{F}(\mathbf{z}, \mathbf{x}, \mathbf{u}) \\ \mathbf{G}(\mathbf{z}, \mathbf{x}, \mathbf{u}) \end{bmatrix} = \mathbf{0}$$

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Semi-explicit DAE:

$$\tilde{\mathbf{F}} = \left[\begin{array}{c} \dot{\mathbf{x}} - \mathbf{F}\left(\mathbf{z}, \mathbf{x}, \mathbf{u}\right) \\ \mathbf{G}\left(\mathbf{z}, \mathbf{x}, \mathbf{u}\right) \end{array} \right] = \mathbf{0}$$

Then the matrix:

$$\begin{bmatrix} \frac{\partial \tilde{\mathbf{F}}}{\partial \dot{\mathbf{x}}} & \frac{\partial \tilde{\mathbf{F}}}{\partial \mathbf{z}} \end{bmatrix} = \begin{bmatrix} I & \frac{\partial \tilde{\mathbf{F}}}{\partial \mathbf{z}} \\ 0 & \frac{\partial \mathbf{G}}{\partial \mathbf{z}} \end{bmatrix}$$

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is full-rank if $\frac{\partial \mathbf{G}}{\partial \mathbf{z}}$ is full rank

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$$\mathbf{F}(\dot{\mathbf{x}},\mathbf{x},\mathbf{z}) = \begin{bmatrix} \mathbf{x} - \dot{\mathbf{x}} + 1\\ \dot{\mathbf{x}}\mathbf{z} + 2 \end{bmatrix} = \mathbf{0}$$

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Then:

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$$\dot{\mathbf{x}} = \mathbf{x} + 1$$
$$\mathbf{z} = -\frac{2}{\dot{\mathbf{x}}} = -\frac{2}{\mathbf{x} + 1}$$

I.e. we can compute $\dot{\mathbf{x}},\,\mathbf{z}$ from \mathbf{x}

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$$\mathbf{F}\left(\dot{\mathbf{x}},\mathbf{x},\mathbf{z}\right) = \left[\begin{array}{c} \dot{\mathbf{x}}_{1} - \mathbf{z} \\ \dot{\mathbf{x}}_{2} - \mathbf{x}_{1} \\ \mathbf{x}_{2} - \mathbf{u} \end{array} \right] = \mathbf{0}$$

Then:

$$\begin{bmatrix} \frac{\partial \mathbf{F}}{\partial \dot{\mathbf{x}}_{1,2}} & \frac{\partial \mathbf{F}}{\partial \mathbf{z}} \end{bmatrix} = \begin{bmatrix} 1 & 0 & -1 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{bmatrix}$$

is rank-deficient.

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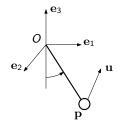
Then:

$$\left[\begin{array}{cc} \frac{\partial \mathbf{F}}{\partial \dot{\mathbf{x}}_{1,2}} & \frac{\partial \mathbf{F}}{\partial \mathbf{z}} \end{array}\right] = \left[\begin{array}{ccc} 1 & 0 & -1 \\ 0 & 1 & 0 \\ 0 & 0 & 0 \end{array}\right]$$

is rank-deficient. This is a "not easy" DAE !! We cannot write $\dot{\mathbf{x}}_{1,2}$ and \mathbf{z} as functions of $\mathbf{x}_{1,2}...$

Model is a semi-explicit DAE with $\mathbf{x} = \begin{bmatrix} \mathbf{p} \\ \mathbf{v} \end{bmatrix}$

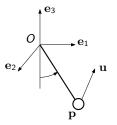
$$\begin{bmatrix} \dot{\mathbf{p}} \\ \dot{\mathbf{v}} \end{bmatrix} = \dot{\mathbf{x}} = \underbrace{\begin{bmatrix} \mathbf{v} \\ \mathbf{v} \\ \frac{\mathbf{u}}{m} - g\mathbf{e}_3 - \frac{z}{m}\mathbf{p} \end{bmatrix}}_{\mathbf{G}(\mathbf{x})}$$



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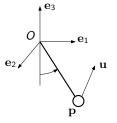
Consider the root-finding problem to be solved in $\dot{\mathbf{x}}, z$:

$$\mathbf{r}(\dot{\mathbf{x}},\mathbf{x},z,\mathbf{u}) = \begin{bmatrix} \dot{\mathbf{x}} - \mathbf{F}(\mathbf{x},z,\mathbf{u}) \\ \mathbf{G}(\mathbf{x}) \end{bmatrix} = \mathbf{0}$$

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Model is a **semi-explicit** DAE with $\mathbf{x} = \begin{bmatrix} \mathbf{p} \\ \mathbf{y} \end{bmatrix}$

$$\begin{bmatrix} \dot{\mathbf{p}} \\ \dot{\mathbf{v}} \end{bmatrix} = \dot{\mathbf{x}} = \underbrace{\begin{bmatrix} \mathbf{v} \\ \mathbf{w} \\ \frac{\mathbf{u}}{m} - g\mathbf{e}_3 - \frac{z}{m}\mathbf{p} \end{bmatrix}}_{\mathbf{G}(\mathbf{x})}$$



Consider the root-finding problem to be solved in $\dot{\mathbf{x}}, z$:

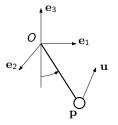
$$\mathbf{r}(\dot{\mathbf{x}},\mathbf{x},z,\mathbf{u}) = \begin{bmatrix} \dot{\mathbf{x}} - \mathbf{F}(\mathbf{x},z,\mathbf{u}) \\ \mathbf{G}(\mathbf{x}) \end{bmatrix} = 0$$

 $\begin{array}{c} \text{Then:} \\ \nabla_{\dot{\mathbf{x}},z}\mathbf{r}^{\top} = \left[\begin{array}{ccc} \textit{I} & 0 & 0 \\ 0 & \textit{I} & \mathbf{p} \\ 0 & 0 & 0 \end{array} \right] \text{ is rank-deficient. The Newton step does not exist !!} \end{array}$

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Model is a semi-explicit DAE with $\mathbf{x} = \begin{vmatrix} \mathbf{p} \\ \mathbf{v} \end{vmatrix}$

$$\begin{bmatrix} \dot{\mathbf{p}} \\ \dot{\mathbf{v}} \end{bmatrix} = \dot{\mathbf{x}} = \underbrace{\begin{bmatrix} \mathbf{v} \\ \frac{\mathbf{u}}{m} - g\mathbf{e}_3 - \frac{z}{m}\mathbf{p} \end{bmatrix}}_{\mathbf{0} = \underbrace{\mathbf{p}}^{\top}\mathbf{p} - L^2}_{\mathbf{G}(\mathbf{x})}}$$



Note that $\frac{\partial \mathbf{G}(\mathbf{x})}{\partial z} = 0$!!

Consider the root-finding problem to be solved in $\dot{\mathbf{x}}, z$:

$$\mathbf{r}(\dot{\mathbf{x}},\mathbf{x},z,\mathbf{u}) = \begin{bmatrix} \dot{\mathbf{x}} - \mathbf{F}(\mathbf{x},z,\mathbf{u}) \\ \mathbf{G}(\mathbf{x}) \end{bmatrix} = \mathbf{0}$$

Then: $\nabla_{\dot{\mathbf{x}},z} \mathbf{r}^{\top} = \begin{bmatrix} I & 0 & 0 \\ 0 & I & \mathbf{p} \\ 0 & 0 & 0 \end{bmatrix}$ is rank-deficient. The Newton step does not exist !!

DAE - Delta Robot

Lagrange model yields a semi-explicit DAE with:

$$\mathbf{G}(\mathbf{x}) = \begin{bmatrix} \|\mathbf{p} - \mathbf{p}_1\|^2 - L^2 \\ \|\mathbf{p} - \mathbf{p}_2\|^2 - L^2 \\ \|\mathbf{p} - \mathbf{p}_3\|^2 - L^2 \end{bmatrix}$$

where

$$\mathbf{p}_{k} = R_{k}^{z} = \begin{bmatrix} \cos \gamma_{k} & \sin \gamma_{k} & 0\\ -\sin \gamma_{k} & \cos \gamma_{k} & 0\\ 0 & 0 & 1 \end{bmatrix} \begin{bmatrix} L \cos \alpha_{k} \\ 0\\ L \sin \alpha_{k} \end{bmatrix}$$

using $\gamma_{1,2,3} = \{0, \frac{2\pi}{3}, \frac{4\pi}{3}\}.$



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using
$$\gamma_{1,2,3} = \{0, \frac{2\pi}{3}, \frac{4\pi}{3}\}.$$

Algebraic variables \mathbf{z} for the forces in the arms:

$$\frac{\partial \mathbf{G}\left(\mathbf{x}\right)}{\partial \mathbf{z}} = \mathbf{0}$$

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using
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Algebraic variables \mathbf{z} for the forces in the arms:

$$\frac{\partial \mathbf{G}\left(\mathbf{x}\right)}{\partial \mathbf{z}} = \mathbf{0}$$

Such that the DAE:

$$\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, z, \mathbf{u})$$
$$\mathbf{0} = \mathbf{G}(\mathbf{x})$$

... cannot be solved for z, because $\frac{\partial \mathbf{G}(\mathbf{x})}{\partial \mathbf{z}} = 0$

Optimal Control with DAEs, lecture 1



Is that a general problem in Lagrange mechanics ? Pretty much ...

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Is that a general problem in Lagrange mechanics ? Pretty much ...

The difficulty comes from having holonomic (aka purely position-dependent) constraints:

 $\mathbf{G}\left(\mathbf{q}\right)=0$

which "hold the system together".

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Is that a general problem in Lagrange mechanics ? Pretty much ...

The difficulty comes from having holonomic (aka purely position-dependent) constraints:

$$\mathbf{G}(\mathbf{q}) = \mathbf{0}$$

which "hold the system together". Then the Euler-Lagrange equations:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial\mathcal{L}}{\partial\dot{\mathbf{q}}} - \frac{\partial\mathcal{L}}{\partial\mathbf{q}} = \mathbf{0}$$

deliver an explicit ODE for the accelerations $\ddot{\mathbf{q}}$, involving the algebraic variables \mathbf{z} .

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$$\mathbf{G}(\mathbf{q}) = \mathbf{0}$$

which "hold the system together". Then the Euler-Lagrange equations:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{0}$$

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What is going on ?!?

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Outline

1 "Easy" & "Hard" DAEs



Index Reduction

Constraints drift

Optimal Control with DAEs, lecture 11

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Definition:

The DAE **differential index** is the minimum *i* such that:

$$\frac{\mathrm{d}^{i}}{\mathrm{d}t^{i}}\mathbf{F}\left(\dot{\mathbf{x}},\mathbf{x},\mathbf{z},\mathbf{u}\right)=\mathbf{0}$$

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Using (to write a $1^{\rm st}\text{-order ODE})$

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Image: A matrix

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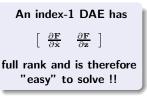
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For a semi-explicit DAE the differential index is the minimum *i* such that:

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Remark: for an index-1 semi-explicit DAE:

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S. Gros

For a **semi-explicit DAE** the differential index is the minimum *i* such that:

$$\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$$
$$0 = \frac{\mathrm{d}^{i}}{\mathrm{d}t^{i}} \mathbf{G}(\mathbf{x}, \mathbf{z}, \mathbf{u})$$

is an ODE

Example:

$$\begin{bmatrix} \dot{\mathbf{x}}_1 \\ \dot{\mathbf{x}}_2 \end{bmatrix} = \begin{bmatrix} 0 & 1 \\ 0 & 0 \end{bmatrix} \begin{bmatrix} \mathbf{x}_1 \\ \mathbf{x}_2 \end{bmatrix} + \begin{bmatrix} 0 \\ 1 \end{bmatrix} z$$
$$0 = \underbrace{\frac{1}{2} \left(\mathbf{x}_1^2 + \mathbf{x}_2^2 - 1 \right)}_{\mathbf{G}(\mathbf{x})}$$

Then $\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{G} = \mathbf{x}_1\mathbf{x}_2 + \mathbf{x}_2z = \mathbf{0}$ $\frac{\mathrm{d}^2}{\mathrm{d}t^2}\mathbf{G} = \dot{\mathbf{x}}_1\mathbf{x}_2 + \mathbf{x}_1\dot{\mathbf{x}}_2 + \dot{\mathbf{x}}_2z + \mathbf{x}_2\dot{z} = \mathbf{0}$

Remark: for an index-1 semi-explicit DAE:

$$\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{G}(\mathbf{x},\mathbf{z},\mathbf{u}) = \frac{\partial \mathbf{G}}{\partial \mathbf{x}}\mathbf{F} + \frac{\partial \mathbf{G}}{\partial \mathbf{z}}\dot{\mathbf{z}} + \frac{\partial \mathbf{G}}{\partial \mathbf{u}}\dot{\mathbf{u}} = \mathbf{0}$$

yields a pure ODE. We have:

$$\dot{\mathbf{z}} = -\frac{\partial \mathbf{G}}{\partial \mathbf{z}}^{-1} \left(\frac{\partial \mathbf{G}}{\partial \mathbf{x}} \mathbf{F} + \frac{\partial \mathbf{G}}{\partial \mathbf{u}} \dot{\mathbf{u}} \right)$$

such that $\frac{\partial \mathbf{G}}{\partial \mathbf{z}}$ is full rank !!

Example is an index-2 DAE

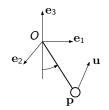
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Example: 3D pendulum

$$m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
$$\mathbf{0} = \underbrace{\frac{1}{2}\left(\mathbf{p}^{\top}\mathbf{p} - \boldsymbol{L}^2\right)}_{\mathbf{G}(\mathbf{x})}$$

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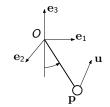
Example: 3D pendulum

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Perform two time differentiations on ${\bf G}$ yields:

$$\ddot{\mathbf{G}} = \frac{1}{2} \left(\ddot{\mathbf{p}}^{\top} \mathbf{p} + \mathbf{p}^{\top} \ddot{\mathbf{p}} + 2 \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} \right) = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} = 0$$

$$egin{aligned} \dot{\mathbf{x}} &= \mathbf{F}\left(\mathbf{x},\mathbf{z},\mathbf{u}
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ight) \ \mathbf{s} ext{ an ODE} \end{aligned}$$



Example: 3D pendulum

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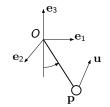
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Substitute $\ddot{\mathbf{p}}$ from $m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$ yields:

$$\mathbf{p}^{\top}\left(\frac{1}{m}\mathbf{u}-g\mathbf{e}_{3}-\frac{1}{m}z\mathbf{p}\right)+\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}}=\mathbf{0}$$

$$\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$$
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Example: 3D pendulum

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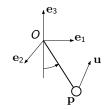
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Substitute $\ddot{\mathbf{p}}$ from $m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$ yields:

$$\mathbf{p}^{\top} \left(\frac{1}{m} \mathbf{u} - g \mathbf{e}_3 - \frac{1}{m} z \mathbf{p} \right) + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} = \mathbf{0}$$

i.e.
$$z = \frac{1}{\mathbf{p}^{\top} \mathbf{p}} \left(\mathbf{p}^{\top} \mathbf{u} - mg \mathbf{p}^{\top} \mathbf{e}_3 + m \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} \right)$$

$$\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$$
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Example: 3D pendulum

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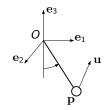
i.e. $z = \frac{1}{\mathbf{p}^{\top}\mathbf{p}} \left(\mathbf{p}^{\top}\mathbf{u} - mg\mathbf{p}^{\top}\mathbf{e}_{3} + m\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \right)$

A third time differentiation yields an ODE for z:

$$\dot{z} = \frac{\mathrm{d}}{\mathrm{d}t} \left[\frac{1}{\mathbf{p}^{\top} \mathbf{p}} \left(\mathbf{p}^{\top} \mathbf{u} - mg \mathbf{p}^{\top} \mathbf{e}_{3} + m\dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} \right) \right]$$

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Example: 3D pendulum

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Perform $\ensuremath{\text{two}}$ time differentiations on $\ensuremath{\mathbf{G}}$ yields:

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Substitute $\ddot{\mathbf{p}}$ from $m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$ yields:

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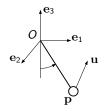
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The 3D pendulum in Lagrange is an index-3 DAE !!

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22nd of February, 2016

S. Gros

Optimal Control with DAEs, lecture 11

Example: 3D pendulum

$$m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
$$0 = \underbrace{\frac{1}{2}\left(\mathbf{p}^{\top}\mathbf{p} - \boldsymbol{L}^2\right)}_{\mathbf{G}(\mathbf{x})}$$

Perform two time differentiations on ${\bf G}$ yields:

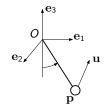
$$\ddot{\mathbf{G}} = \frac{1}{2} \left(\ddot{\mathbf{p}}^{\top} \mathbf{p} + \mathbf{p}^{\top} \ddot{\mathbf{p}} + 2 \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} \right) = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} = \mathbf{0}$$

Assemble:

$$m\ddot{\mathbf{p}} + z\mathbf{p} = \mathbf{u} - mg\mathbf{e}_3$$

 $\mathbf{p}^{\top}\ddot{\mathbf{p}} = -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}}$

$$\dot{\mathbf{x}} = \mathbf{F}\left(\mathbf{x}, \mathbf{z}, \mathbf{u}
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Example: 3D pendulum

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Assemble:

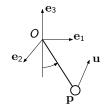
$$m\ddot{\mathbf{p}} + z\mathbf{p} = \mathbf{u} - mg\mathbf{e}_{z}$$

 $\mathbf{p}^{ op}\ddot{\mathbf{p}} = -\dot{\mathbf{p}}^{ op}\dot{\mathbf{p}}$

in matrix form yields:

$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

$$\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$$
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$$m\ddot{\mathbf{p}} + z\mathbf{p} = \mathbf{u} - mg\mathbf{e}_3$$

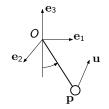
 $\mathbf{p}^{\top}\ddot{\mathbf{p}} = -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}}$

in matrix form yields:

$$\begin{bmatrix} mI & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

This is an index-1 (i.e. "easy") DAE !!

$$egin{aligned} \dot{\mathbf{x}} &= \mathbf{F}\left(\mathbf{x}, \mathbf{z}, \mathbf{u}
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Assemble:

$$egin{aligned} m\ddot{\mathbf{p}}+z\mathbf{p}&=\mathbf{u}-mg\mathbf{e}_3\ \mathbf{p}^{ op}\ddot{\mathbf{p}}&=-\dot{\mathbf{p}}^{ op}\dot{\mathbf{p}} \end{aligned}$$

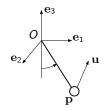
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We have converted the index-3 DAE into an index-1 DAE !!

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Example: 3D pendulum

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Perform $\ensuremath{\text{two}}$ time differentiations on $\ensuremath{\mathbf{G}}$ yields:

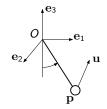
$$\ddot{\mathbf{G}} = \frac{1}{2} \left(\ddot{\mathbf{p}}^{\top} \mathbf{p} + \mathbf{p}^{\top} \ddot{\mathbf{p}} + 2 \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} \right) = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}} = \mathbf{0}$$

Transforming a high-index DAE into an equivalent lower-index one is labelled **index reduction**

For a **semi-explicit DAE** the differential index is the minimum *i* such that:

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s an ODE



We have converted the index-3 DAE into an index-1 DAE !!

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Outline

1 "Easy" & "Hard" DAEs

Differential Index



Constraints drift

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DAEs from Lagrange Mechanics

Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
$$\mathbf{c}\left(\mathbf{q}\right) = \mathbf{0}$$

with $\mathcal{L}(\mathbf{q}, \dot{\mathbf{q}}, \mathbf{z}) = \mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - \mathbf{V}(\mathbf{q}) - \mathbf{z}^{\top} \mathbf{c}(\mathbf{q})$

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DAEs from Lagrange Mechanics

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For most mechanical applications:

$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}\mathcal{M}(\mathbf{q})\dot{\mathbf{q}}$$

such that:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}}^{\top} = M(\mathbf{q})\ddot{\mathbf{q}} + \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}}$$

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DAEs from Lagrange Mechanics

Index-3 DAE from Lagrange:

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Then the differential part of the DAE model reads as:

 $M(\mathbf{q})\ddot{\mathbf{q}} + \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} - \nabla_{\mathbf{q}}\left(\mathcal{T}\left(\mathbf{q}, \dot{\mathbf{q}}\right) - \mathcal{V}\left(\mathbf{q}\right)\right) + \nabla \mathbf{c}\left(\mathbf{q}\right)\mathbf{z} = \mathbf{F}_{g}$

For most mechanical applications:

$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}\mathcal{M}(\mathbf{q})\dot{\mathbf{q}}$$

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$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial\mathcal{L}}{\partial\dot{\mathbf{q}}}^{\top} = M(\mathbf{q})\ddot{\mathbf{q}} + \dot{M}(\mathbf{q},\dot{\mathbf{q}})\dot{\mathbf{q}}$$

Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
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with $\mathcal{L}(\mathbf{q}, \dot{\mathbf{q}}, \mathbf{z}) = \mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - \mathbf{V}(\mathbf{q}) - \mathbf{z}^{\top} \mathbf{c}(\mathbf{q})$

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The 1^{st} and 2^{nd} -order time derivatives of c(q) read as:

$$\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}},$$

For most mechanical applications:

$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}\mathcal{M}(\mathbf{q})\dot{\mathbf{q}}$$

such that:

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Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
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abla_{\mathbf{q}}\left(\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight) - V\left(\mathbf{q}
ight)
ight) +
abla_{\mathbf{c}}\left(\mathbf{q}
ight)\mathbf{z} = \mathbf{F}_{g}$$

The $1^{\rm st}$ and $2^{\rm nd}\text{-order}$ time derivatives of $c\left(q\right)$ read as:

$$\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}}, \qquad \frac{\mathrm{d}^{2}}{\mathrm{d}t^{2}}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\ddot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}}\right)^{\top}\dot{\mathbf{q}}$$

 $\langle \Box \rangle \langle \Box \rangle$ 22^{nd} of February, 2016

Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
$$\mathbf{c}(\mathbf{q}) = \mathbf{0}$$

with $\mathcal{L}(\mathbf{q}, \dot{\mathbf{q}}, \mathbf{z}) = \mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - \mathcal{V}(\mathbf{q}) - \mathbf{z}^{\top} \mathbf{c}(\mathbf{q})$

For most mechanical applications:

$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}\mathcal{M}(\mathbf{q})\dot{\mathbf{q}}$$

such that:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}}^{\top} = M(\mathbf{q})\ddot{\mathbf{q}} + \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}}$$

Then the differential part of the DAE model reads as:

$$M(\mathbf{q})\ddot{\mathbf{q}} + \dot{M}(\mathbf{q},\dot{\mathbf{q}})\dot{\mathbf{q}} -
abla_{\mathbf{q}}\left(\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight) - \mathcal{V}\left(\mathbf{q}
ight)
ight) +
abla_{\mathbf{c}}\left(\mathbf{q}
ight)\mathbf{z} = \mathbf{F}_{\mathrm{g}}$$

The 1st and 2nd-order time derivatives of $\mathbf{c}(\mathbf{q})$ read as: $\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}(\mathbf{q}) = \nabla \mathbf{c}(\mathbf{q})^{\top} \dot{\mathbf{q}}, \qquad \frac{\mathrm{d}^{2}}{\mathrm{d}t^{2}}\mathbf{c}(\mathbf{q}) = \nabla \mathbf{c}(\mathbf{q})^{\top} \ddot{\mathbf{q}} + \nabla_{\mathbf{q}} \left(\nabla \mathbf{c}(\mathbf{q})^{\top} \dot{\mathbf{q}}\right)^{\top} \dot{\mathbf{q}}$

Index-1 DAE model:

$$\begin{bmatrix} M(\mathbf{q}) & \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q}) \\ \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q})^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{q}} \\ \mathbf{z} \end{bmatrix} = \begin{bmatrix} \mathbf{F}_{g} - \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - V(\mathbf{q})\right) \\ -\nabla_{\mathbf{q}}\left(\nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q})^{\top} \dot{\mathbf{q}}\right)^{\top} \dot{\mathbf{q}} \end{bmatrix}$$

Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
$$\mathbf{c}(\mathbf{q}) = \mathbf{0}$$

with $\mathcal{L}(\mathbf{q}, \dot{\mathbf{q}}, \mathbf{z}) = \mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - \mathbf{V}(\mathbf{q}) - \mathbf{z}^{\top} \mathbf{c}(\mathbf{q})$

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$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
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Index-1 DAE model:

$$\begin{bmatrix} M(\mathbf{q}) & \nabla \mathbf{c}(\mathbf{q}) \\ \nabla \mathbf{c}(\mathbf{q})^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{q}} \\ \mathbf{z} \end{bmatrix} = \begin{bmatrix} \mathbf{F}_{g} - \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - V\left(\mathbf{q}\right)\right) \\ -\nabla_{\mathbf{q}}\left(\nabla \mathbf{c}\left(\mathbf{q}\right)^{\top} \dot{\mathbf{q}}\right)^{\top} \dot{\mathbf{q}} \end{bmatrix}$$

Models based on Lagrange mechanics typically are **index-3 DAEs**, making them intrinsically difficult to use. The best approach to treat them is usually to proceed with an **index reduction** down to **index 1** for which very classical integration tools work well.

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High-index semi-explicit DAE

 $\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$ $\mathbf{0} = \mathbf{G}(\mathbf{x}, \mathbf{z}, \mathbf{u})$

Algorithm (see "Nonlinear Programming", L.T. Biegler)

- 2

High-index semi-explicit DAE

 $\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u})$ $0 = \mathbf{G}(\mathbf{x}, \mathbf{z}, \mathbf{u})$

Algorithm (see "Nonlinear Programming", L.T. Biegler)

- 1 Check if the DAE system is index 1 (i.e. $\frac{\partial G}{\partial \sigma}$ full rank). If yes, stop.
- Identify a subset of algebraic equations that can be solved for a subset of algebraic variables.
- 3 Apply $\frac{d}{dt}$ on the remaining algebraic equations that contain the differential variables x_i .
- **4** Terms $\dot{\mathbf{x}}_i$ will appear in these differentiated equations.
- Substitute the $\dot{\mathbf{x}}_i$ with $\mathbf{F}_i(\mathbf{x}, \mathbf{z}, \mathbf{u})$. This leads to new algebraic equations.
- With this new DAE system, go to step 1.

High-index semi-explicit DAE

 $\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u}) \\ \mathbf{0} = \mathbf{G}(\mathbf{x}, \mathbf{z}, \mathbf{u})$

Algorithm (see "Nonlinear Programming", L.T. Biegler)

- Check if the DAE system is index 1 (i.e. $\frac{\partial G}{\partial z}$ full rank). If yes, stop.
- Identify a subset of algebraic equations that can be solved for a subset of algebraic variables.
- Apply d/dt on the remaining algebraic equations that contain the differential variables x_j.
- **(4)** Terms $\dot{\mathbf{x}}_j$ will appear in these differentiated equations.
- Substitute the $\dot{\mathbf{x}}_j$ with $\mathbf{F}_j(\mathbf{x}, \mathbf{z}, \mathbf{u})$. This leads to new algebraic equations.
- **(5)** With this new DAE system, go to step 1.

High-index semi-explicit DAE

 $\dot{\mathbf{x}} = \mathbf{F}(\mathbf{x}, \mathbf{z}, \mathbf{u}) \\ \mathbf{0} = \mathbf{G}(\mathbf{x}, \mathbf{z}, \mathbf{u})$

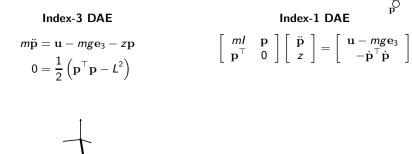
Algorithm (see "Nonlinear Programming", L.T. Biegler)

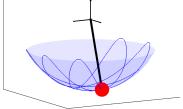
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- Substitute the $\dot{\mathbf{x}}_j$ with $\mathbf{F}_j(\mathbf{x}, \mathbf{z}, \mathbf{u})$. This leads to new algebraic equations.
- With this new DAE system, go to step 1.

Writing a general-purpose "Index-reduction algorithm" can be very tricky, as one of the steps is not easily automated

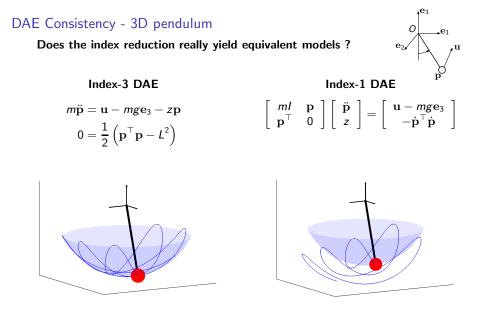
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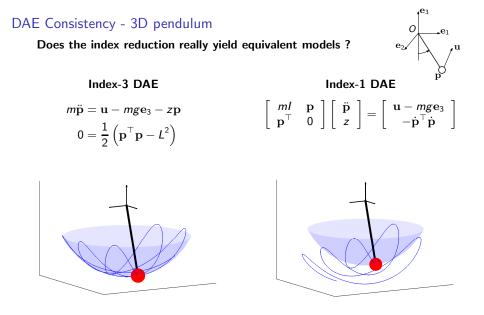
Does the index reduction really yield equivalent models ?





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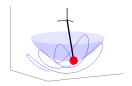


What is going on ??

Optimal Control with DAEs, lecture 11

Index-3 DAE

$$m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
$$\mathbf{0} = \frac{1}{2}\left(\mathbf{p}^{\top}\mathbf{p} - \mathcal{L}^2\right)$$



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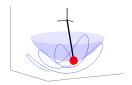
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Index-3 DAE

$$egin{aligned} & m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p} \ & \mathbf{0} = rac{1}{2}\left(\mathbf{p}^ op \mathbf{p} - \mathcal{L}^2
ight) \end{aligned}$$

Index reduction

$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \mathcal{L}^2 \right)$$
$$\dot{\mathbf{c}} = \mathbf{p}^{\top} \dot{\mathbf{p}}$$
$$\ddot{\mathbf{c}} = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}}$$



3

Index-3 DAE

$$m\mathbf{\hat{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
$$\mathbf{0} = \frac{1}{2}\left(\mathbf{p}^{\top}\mathbf{p} - L^2\right)$$

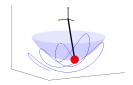
Index reduction

$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \boldsymbol{L}^2 \right)$$
$$\dot{\mathbf{c}} = \mathbf{p}^{\top} \dot{\mathbf{p}}$$
$$\ddot{\mathbf{c}} = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}}$$

Index-1 DAE

$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

... is built to impose $\ddot{\mathbf{c}}=\mathbf{0}$ at all time.



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Index reduction

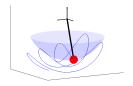
$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \mathcal{L}^2 \right)$$
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... is built to impose $\ddot{\mathbf{c}}=\mathbf{0}$ at all time. But it does not ensure

 $\dot{\mathbf{c}} = \mathbf{0}$ and $\mathbf{c} = \mathbf{0} \parallel$



3

Index-3 DAE

$$m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
$$\mathbf{0} = \frac{1}{2}\left(\mathbf{p}^{\top}\mathbf{p} - L^2\right)$$

Index reduction

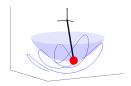
$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \mathcal{L}^2 \right)$$
$$\dot{\mathbf{c}} = \mathbf{p}^{\top} \dot{\mathbf{p}}$$
$$\ddot{\mathbf{c}} = \mathbf{p}^{\top} \ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top} \dot{\mathbf{p}}$$

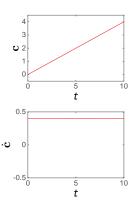
Index-1 DAE

$$\begin{bmatrix} mI & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

... is built to impose $\ddot{\mathbf{c}}=\mathbf{0}$ at all time. But it does not ensure

 $\dot{\mathbf{c}} = \mathbf{0}$ and $\mathbf{c} = \mathbf{0} \ \mathbf{!!}$





Index-3 DAE

$$m\ddot{\mathbf{p}} = \mathbf{u} - mg\mathbf{e}_3 - z\mathbf{p}$$
 $0 = \frac{1}{2} \left(\mathbf{p}^{\top}\mathbf{p} - L^2 \right)$

Index reduction

$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \mathcal{L}^2 \right)$$
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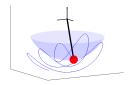
Index-1 DAE

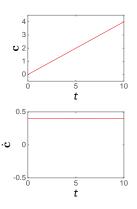
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... is built to impose $\ddot{\mathbf{c}}=\mathbf{0}$ at all time. But it does not ensure

 $\dot{\mathbf{c}} = \mathbf{0}$ and $\mathbf{c} = \mathbf{0} \parallel$

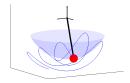
How can we address that ??

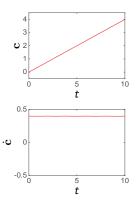




$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

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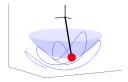
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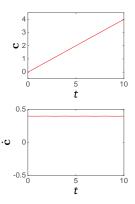
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$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

... is built to impose $\ddot{\mathbf{c}} = 0$ at all time. Then if $\mathbf{c} = 0$ and $\dot{\mathbf{c}} = 0$ are satisfied at **any** time on the trajectory, then they are satisfied at **all** time.



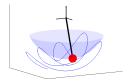


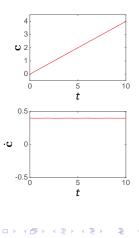
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An index-reduced DAE \underline{must} come with **consistency conditions**. E.g. for the 3D pendulum, the index-1 DAE should be given as:

$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$





$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

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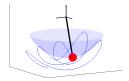
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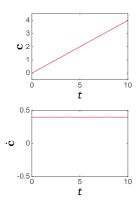
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with the consistency conditions:

$$\mathbf{c} = \frac{1}{2} \left(\mathbf{p}^{\top} \mathbf{p} - \boldsymbol{L}^2 \right) = \mathbf{0}, \qquad \dot{\mathbf{c}} = \mathbf{p}^{\top} \dot{\mathbf{p}} = \mathbf{0}$$

... to be satisfied e.g. at t_0 .





22nd of February, 2016

$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

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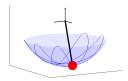
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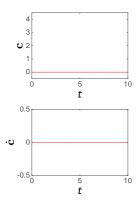
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... to be satisfied e.g. at t_0 .





Consistency of DAEs from Lagrange Mechanics

Index-3 DAE from Lagrange:

$$\frac{\mathrm{d}}{\mathrm{d}}\frac{\partial \mathcal{L}}{\partial \dot{\mathbf{q}}} - \frac{\partial \mathcal{L}}{\partial \mathbf{q}} = \mathbf{F}_{\mathrm{g}}$$
$$\mathbf{c}\left(\mathbf{q}\right) = \mathbf{0}$$

For most mechanical applications:

$$\mathcal{T}\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}\mathcal{M}(\mathbf{q})\dot{\mathbf{q}}$$

with $\mathcal{L}\left(\mathbf{q}, \dot{\mathbf{q}}, \mathbf{z}\right) = \mathcal{T}\left(\mathbf{q}, \dot{\mathbf{q}}\right) - \mathcal{V}\left(\mathbf{q}\right) - \mathbf{z}^{\top}\mathbf{c}\left(\mathbf{q}\right)$

Index reduction based on:

$$\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}} \quad \text{and} \quad \frac{\mathrm{d}^{2}}{\mathrm{d}t^{2}}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\ddot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}}\right)^{\top}\dot{\mathbf{q}}$$

Index-1 DAE model:

$$\begin{bmatrix} M(\mathbf{q}) & \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q}) \\ \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q})^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{q}} \\ \mathbf{z} \end{bmatrix} = \begin{bmatrix} \mathbf{F}_{g} - \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\mathcal{T}(\mathbf{q}, \dot{\mathbf{q}}) - V(\mathbf{q})\right) \\ -\nabla_{\mathbf{q}}\left(\nabla \mathbf{c}(\mathbf{q})^{\top} \dot{\mathbf{q}}\right)^{\top} \dot{\mathbf{q}} \end{bmatrix}$$

Consistency of DAEs from Lagrange Mechanics

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For most mechanical applications:

$$T\left(\mathbf{q},\dot{\mathbf{q}}
ight)=rac{1}{2}\dot{\mathbf{q}}^{ op}M(\mathbf{q})\dot{\mathbf{q}}$$

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$$\frac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}} \quad \text{and} \quad \frac{\mathrm{d}^{2}}{\mathrm{d}t^{2}}\mathbf{c}\left(\mathbf{q}\right) = \nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\ddot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\nabla\mathbf{c}\left(\mathbf{q}\right)^{\top}\dot{\mathbf{q}}\right)^{\top}\dot{\mathbf{q}}$$

Index-1 DAE model:

$$\begin{bmatrix} M(\mathbf{q}) & \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q}) \\ \nabla_{\mathbf{q}}\mathbf{c}(\mathbf{q})^{\top} & 0 \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{q}} \\ \mathbf{z} \end{bmatrix} = \begin{bmatrix} \mathbf{F}_{g} - \dot{M}(\mathbf{q}, \dot{\mathbf{q}})\dot{\mathbf{q}} + \nabla_{\mathbf{q}}\left(\mathcal{T}\left(\mathbf{q}, \dot{\mathbf{q}}\right) - V\left(\mathbf{q}\right)\right) \\ -\nabla_{\mathbf{q}}\left(\nabla\mathbf{c}\left(\mathbf{q}\right)^{\top} \dot{\mathbf{q}}\right)^{\top} \dot{\mathbf{q}} \end{bmatrix}$$

with the consistency conditions:

$$\mathbf{c}\left(\mathbf{q}
ight)=\mathbf{0}$$
 and $rac{\mathrm{d}}{\mathrm{d}t}\mathbf{c}\left(\mathbf{q}
ight)=
abla\mathbf{c}\left(\mathbf{q}
ight)^{ op}\dot{\mathbf{q}}$

Outline

"Easy" & "Hard" DAEs

Differential Index

Index Reduction

4 Constraints drift

Optimal Control with DAEs, lecture 11

 22^{nd} of February, 2016

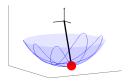
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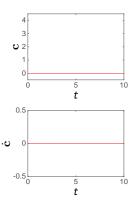
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Index-1 DAE

$$\begin{bmatrix} ml & \mathbf{p} \\ \mathbf{p}^{\top} & \mathbf{0} \end{bmatrix} \begin{bmatrix} \ddot{\mathbf{p}} \\ z \end{bmatrix} = \begin{bmatrix} \mathbf{u} - mg\mathbf{e}_3 \\ -\dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} \end{bmatrix}$$

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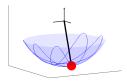


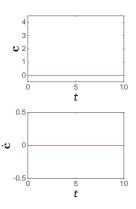
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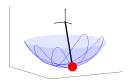
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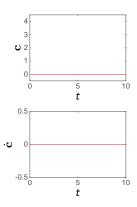
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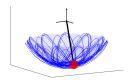
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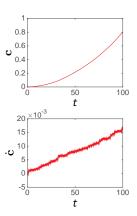
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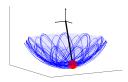
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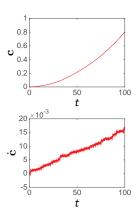
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... imposed at e.g. t_0

With consistent initial conditions, ${\bf c}=0$ and $\dot{{\bf c}}=0$ would be satisfied at all time if we had no numerical error in the integration !!





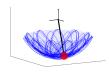
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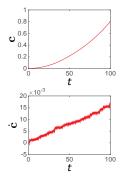
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Why does the drift happen: $\ddot{\mathbf{c}} = 0$ has marginally stable dynamics (\mathbf{c} is two integrations of $\ddot{\mathbf{c}}$ hence two poles at 0).





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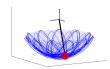
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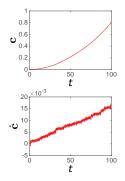
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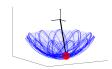
with the consistency conditions:

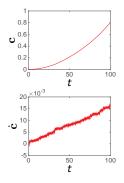
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$$\ddot{\mathbf{c}} + \gamma_1 \dot{\mathbf{c}} + \gamma_0 \mathbf{c} = \mathbf{0}$$





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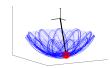
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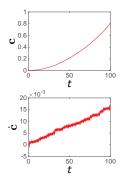
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E.g. for the 3D pendulum:

$$\mathbf{p}^{\top}\ddot{\mathbf{p}} + \dot{\mathbf{p}}^{\top}\dot{\mathbf{p}} + \gamma_{1}\mathbf{p}^{\top}\dot{\mathbf{p}} + \frac{\gamma_{2}}{2}\left(\mathbf{p}^{\top}\mathbf{p} - L^{2}\right) = \mathbf{0}$$





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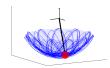
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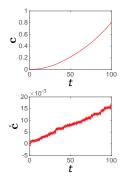
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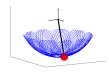
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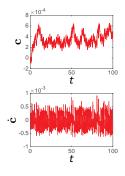
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E.g. poles at -1

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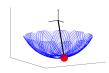
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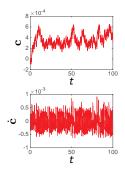
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The Baumgartne stabilization must be used carefully !

- Fast poles introduce stiffness in the dynamics
- The interaction between the stabilization and the integrator error is non-trivial...





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E.g. poles at -1

Consistency & drift are not DAE-specific.

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Simple ODEs

$$\dot{\mathbf{x}} = \mathbf{F}\left(\mathbf{x}, \mathbf{u}\right)$$

model a physical reality. Some ODEs are representative only when some consistency conditions:

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Not covered in this course but good to know: Symplectic integrators (for handling drift)