the wake in pictures

Rachel Leuthold and Moritz Diehl

Albert-Ludwigs-University, Freiburg, Germany





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- ask questions!
- wave at me if I speak too quickly/quietly/etc
- sit at the front



wake:

the disturbed flow (typically turbulent) behind an obstacle

here: ship traveling right-to-left



Newman 1970, in Van Dyke's Album of Fluid Motion (1982, Parabolic, Stanford, CA, USA)



wake:

the disturbed flow (typically turbulent) behind an obstacle

here:

condensation in Horns Rev, Denmark





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let's zoom in a bit...



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the disturbed flow (typically turbulent) behind an obstacle

here: condensation in Horns Rev. Denmark



what do we see? the wake is pretty big,

and some of the turbines are sitting directly in it



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farm perspective:

the flow at 'later' turbines is not fresh and won't contain as much energy

wake:

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what do we see? the wake is pretty big,

and some of the turbines are sitting directly in it

farm perspective:

the flow at 'later' turbines is not fresh and won't contain as much energy

individual perspective: wake limits 'extractable' power by up to 60%

and influences local apparent velocity at blades

what's in here?



- what is the wake doing?
 - \rightarrow $\;$ some philosophical implications of lift
 - ightarrow an angular momentum detour
 - \rightarrow some experiments
 - how does the wake affect wind turbines
 - one conceptual way to model the wake
- where to find more practical information

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'circulation', not 'rotation'





circulation around body \Leftrightarrow lift!

























| ۸ | | | |
|---------|---------------------------|----------------|---|
| | circulation bound to wing | | |
| | lift | | |
| | speed | | |
| | | | |
| at rest | impulsive start | impulsive stop | t |





| | angular momentum of flow 'bound' to wing | |
|---------|--|--------|
| Î | circulation bound to wing | |
| | lift | |
| | speed | |
| | | |
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| Î | circulation bound to wing | | | |
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| | | | | |
| at rest | impulsive start | impulsive stop | t | > |

what part of the flow might be acting to conserve angular momentum?

(not to scale!)



wake:

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a hypothesis: maybe the wake acts to conserve angular momentum!



wake:

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a hypothesis: maybe the wake acts to conserve angular momentum! hypothetically, how would this work?



| | | angular momentum of flow 'bound' to wing | | |
|------|------|--|------|----------------------|
| , | | circulation bound to wing | | |
| | | lift | | |
| | | speed | | |
| | | | | |
| at r | rest | impulsive start impulsive | stop | \overrightarrow{t} |



| | | angular momentum of flow 't | oound' to wing | |
|------|---------------------------|-----------------------------|------------------|-------------------|
| , | circulation bound to wing | | | |
| | lift | | | |
| | | speed | | |
| | | | | |
| at r | rest | impulsive start | impulsive stop | $t \rightarrow t$ |
| | | | | |
| | | | | |
| | | | | |
| | | angular momentum sned int | to wake at start | |
| | | (not to sca | le!) | |





| | angular momentum of flow 'bound' to wing | |
|---------|--|-------------------|
| Î | circulation bound to wing | |
| | lift | |
| | speed | |
| | | |
| at rest | impulsive start impulsive stop | \xrightarrow{t} |
| | | |
| | İ | |
| | l Langular momentum 'shed' into wake at start | |
| | angular momentum sned into wake at start | |
| | (not to scale!) | |







a plan of experiments















Tietjens 1934, in Tietjens and Prandtl's Fundamentals of Hydro- and Aeromechanics (1957, tr. Rosenhead, Dover, New York, NY, USA)







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2D experiment: impulsive start is 'anything' shed into wake?







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2D experiment: impulsive start is 'anything' shed into wake? YES!







a plan of experiments



2D experiment: impulsive start is 'anything' shed into wake?

2D experiment: start and stop is circulation shed into wake?



2D experiment: start and stop is circulation shed into wake?







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2D experiment: start and stop is circulation shed into wake?







2D experiment: start and stop is circulation shed into wake? YES!







from infinite to finite wings ...





from infinite to finite wings...





from infinite to finite wings...





is there circulating motion trailing behind wing tips?

a plan of experiments



2D experiment: impulsive start is 'anything' shed into wake?

2D experiment: start and stop is circulation shed into wake?

3D experiment: steady, level flight does circulation trail behind wingtips?



3D experiment: steady, level flight does circulation trail behind wingtips?





Head 1982, in Van Dyke's Album of Fluid Motion (1982, Parabolic, Stanford, CA, USA)

Rachel Leuthold

3D experiment: steady, level flight does circulation trail behind wingtips? YES!





Head 1982, in Van Dyke's Album of Fluid Motion (1982, Parabolic, Stanford, CA, USA)

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from steady level flight to rotors...





Head 1982, in Van Dyke's Album of Fluid Motion (1982, Parabolic, Stanford, CA, USA). Rachel Leuthold

from steady level flight to rotors...





is there circulating motion trailing along the helical path behind the rotating wingtips?

left: in Hand, et al. Unsteady Aerodynamics Experiment Phase VI: Wind tunnel.... (2001, NREL/TP-500-29955, Golden, CO, USA.) right: source unknown.

a plan of experiments



2D experiment: impulsive start is 'anything' shed into wake?

2D experiment: start and stop is circulation shed into wake?

3D experiment: steady, level flight does circulation trail behind wingtips?

3D experiment: steady turbine rotor does circulation trail behind wingtips?



3D experiment: steady turbine rotor does circulation trail behind wingtips?





in Vermeer, 'A review of wind turbine wake research ...' in ASME Wind Energy Symp. Tech. Papers (2001, ASME, New York, NY, USA)

3D experiment: steady turbine rotor does circulation trail behind wingtips?





in Alfredsson and Dahlberg A preliminary wind tunnel study of windmill wake... (1979, FFA/TN-AU-1499 part 7, Stockholm, Sweden)

3D experiment: steady turbine rotor does circulation trail behind wingtips? YES!



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what is the wake doing?



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the wake is conserving angular momentum! (are we convinced?)

Vattenfall 2010, https://www.flickr.com/photos/vattenfall/4270899001

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how does the wake affect the turbine?





(not to scale!)

the wake decreases the velocity...





the wake decreases the velocity at the rotor!



the wake decreases the velocity at the rotor!



remember: $P \propto u^3$!





apparent velocity $oldsymbol{u}_{\mathrm{a}}$











wake 'effectively' increases drag!







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imagine the following...





there is an invisible state conducted in a structure and produces a force field circulates certain particles

imagine the following...





there is an invisible state conducted in a structure and produces a force field circulates certain particles current wire electromagnetic magnetic
imagine the following...





there is an invisible state conducted in a structure and produces a force field circulates certain particles currentvorticitywirevortexelectromagneticpressuremagneticfluid



electromagnetism

Poisson equation for electrostatics:

$$\begin{array}{c} \nabla^2 A = -\mu_0 J\\ A & \text{vector potential}\\ J & \text{current density} \end{array}$$

Ampère's law:

$$\underbrace{ \boldsymbol{J} = \nabla \times \boldsymbol{H} }_{\boldsymbol{H} \quad \text{magnetic field intensity} }$$

aerodynamics

for incompressible flows:

$$abla^2 \psi = -\omega$$

- ψ vector potential
- ω vorticity

definition of vorticity:

 \boldsymbol{u}

$$\underline{\omega}:=
abla imes oldsymbol{u}$$
flow velocity, s.t.

$$oldsymbol{u} = oldsymbol{u}_{\infty} +
abla imes oldsymbol{\psi}$$



electromagnetism

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define 'circulation' as flux of vorticity through surface S $\Gamma := \int_S \, {m \omega} \cdot \hat{m n} \, dS$



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then, Kutta-Joukowski expression says: 'lift per unit span is proportional to circulation' $(L' = \rho u \Gamma)$



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solution:

$$oldsymbol{u}(oldsymbol{x}) = oldsymbol{u}_{\infty} - rac{1}{4\pi} \int_{V} rac{(oldsymbol{x} - oldsymbol{\xi})}{||oldsymbol{x} - oldsymbol{\xi}||_{2}^{3}} imes oldsymbol{\omega}(oldsymbol{\xi}) \,\,doldsymbol{\xi}$$



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solution:

YUCK!
$$\boldsymbol{u}(\boldsymbol{x}) = \boldsymbol{u}_{\infty} - \frac{1}{4\pi} \int_{V} \frac{(\boldsymbol{x} - \boldsymbol{\xi})}{||\boldsymbol{x} - \boldsymbol{\xi}||_{2}^{3}} \times \boldsymbol{\omega}(\boldsymbol{\xi}) \ d\boldsymbol{\xi}$$



solid boundary



infinitely far away

freestream u_{∞}

solid boundary









the 'external flow': shear gradient ≈ 0

 $u(\delta) = \kappa \ u_{\infty}$

the 'boundary layer': shear gradient strong!









* certain conditions apply



(remember: trying to solve
$$oldsymbol{u}(oldsymbol{x}) = oldsymbol{u}_\infty - rac{1}{4\pi}\int_V rac{(oldsymbol{x}-oldsymbol{\xi})}{||oldsymbol{x}-oldsymbol{\xi}||_2^3} imes oldsymbol{\omega}(oldsymbol{\xi}) \; doldsymbol{\xi}$$
)

 \dots separates V into two parts:

boundary layer Ω vorticity $\omega(\xi) \neq 0$ \downarrow local circulating motion of fluid elements due to viscous shear stress external flow Ω' vorticity $\omega(\boldsymbol{\xi}) \approx 0 *$ \downarrow drops out of integral

* for 'potential' external flow \approx incompressible + inviscid + irrotational.

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the boundary layer saves a lot of effort!



Werlé 1974, in Van Dyke's Album of Fluid Motion (1982, Parabolic, Stanford, CA, USA)

the boundary layer saves a lot of effort!



$$oldsymbol{u}(oldsymbol{x}) pprox oldsymbol{u}_{\infty} - rac{1}{4\pi} \left(\int_{\Omega'} oldsymbol{0} \ doldsymbol{\xi} + \int_{\Omega} rac{(oldsymbol{x} - oldsymbol{\xi})}{||oldsymbol{x} - oldsymbol{\xi}||_2^3} imes oldsymbol{\omega}(oldsymbol{\xi}) \ doldsymbol{\xi}
ight)$$

MESCO

the boundary layer saves a lot of effort!



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ight)$$



$$oldsymbol{u}(oldsymbol{x})pproxoldsymbol{u}_{\infty}\!-\!rac{1}{4\pi}\int_{\Omega}rac{(oldsymbol{x}\!-\!oldsymbol{\xi})}{||oldsymbol{x}\!-\!oldsymbol{\xi}||_2^3}\! imes\!oldsymbol{\omega}(oldsymbol{\xi})\,\,doldsymbol{\xi}$$

discretize $\pmb{\omega}$ into vortices* with known analytical integrals... points

curves like line-segments or helices surfaces like rectangles volumes like spheres

...then sum so that given boundary conditions are satisfied!

$$oldsymbol{u}(oldsymbol{x})pproxoldsymbol{u}_{\infty}\!+\!\sum_{i}^{N}oldsymbol{u}_{i}(oldsymbol{x},oldsymbol{\omega}_{i})$$

* overlapping vortices allowed due to linearity of $abla^2$ operator.

Rachel Leuthold

more practical information can be found in... Mawesco



... in order of increasing thoroughness + rigorousness

thank you for your attention!



... any questions?